

# **Design Safety Used in NASA's Human-rated Primary Lithium Batteries**

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**NASA-JSC**

**The 2013 NASA Battery Workshop**

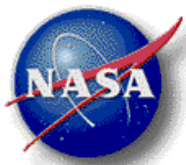
**Huntsville, AL**

**November, 2013**



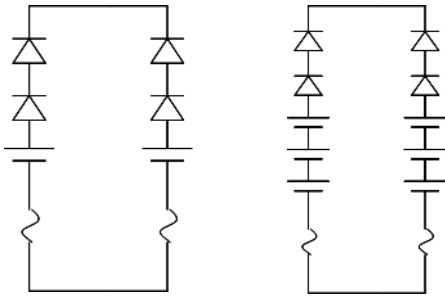
# Safety Requirements

- NASA's human-rated safety requirements
  - Two-failure tolerance to catastrophic failures (JSC 20793)
- Catastrophic for human-rating requires
  - No electrolyte venting that would cause permanent injury to crew or loss of vehicle or mission
  - Electrolytes that can cause permanent injury are rated as Toxicology level 2 for NASA human-rated missions
    - Lithium primary with an organic electrolyte and li-ion with organic electrolytes have a salt that is an irritant and corrosive giving these electrolytes a Tox2 rating
- The requirement therefore dictates that two-failure tolerance should be provided for all hazards that can lead to venting (electrolyte liquid), fire and thermal runaway.
  - For primary batteries this translates to two-failure tolerance to inadvertent charge, overdischarge into reversal, external and internal shorts, extreme temperatures especially high temps
  - Since no external controls exist for internal shorts, a design for minimum risk approach is used to mitigate that hazard.

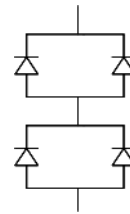


# Recommendations for Protective Features to Prevent Catastrophic Failures

- Inadvertent charge shall be prevented
  - Between parallel primary lithium cells/cell strings
  - Between primary lithium cells and other charging sources

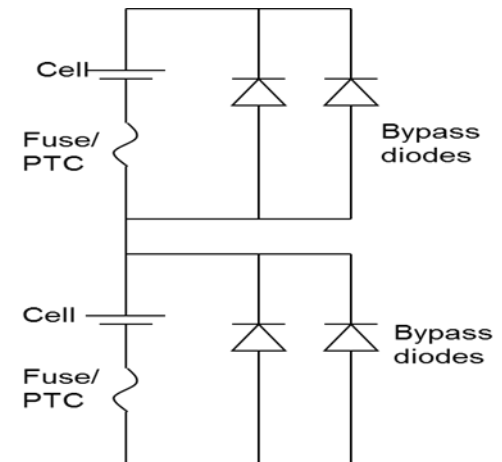


Cells in parallel    Cell strings in parallel



Redundancy in the case of mission critical applications

- Overdischarge into reversal
  - Prevent unbalanced cell condition (need for cell matching)
  - Prevent taking cells into low voltages and/ or reversal

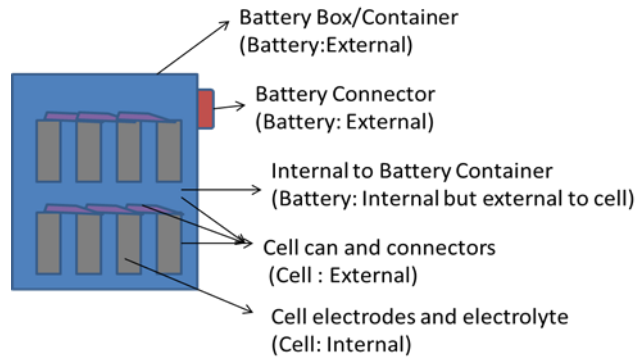




# Recommendations for Protective Features to Prevent Catastrophic Failures

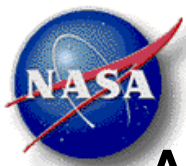
- External Shorts

- High and low impedance short protection required
- Fuses, PTCs, polyswitches, etc. shall be rated for the voltage, current and temperature of battery design and application as well as environment



- Internal Short

- Stringent cell and battery level screening (OCV, CCV, dimensions, mass,  $R_e$ )
- Cell matching
- Rigorous battery flight acceptance testing that includes (vacuum) leak check and vibration screening with stringent pass/fail criteria for OCV, CCV ( $R_e$ ) and mass

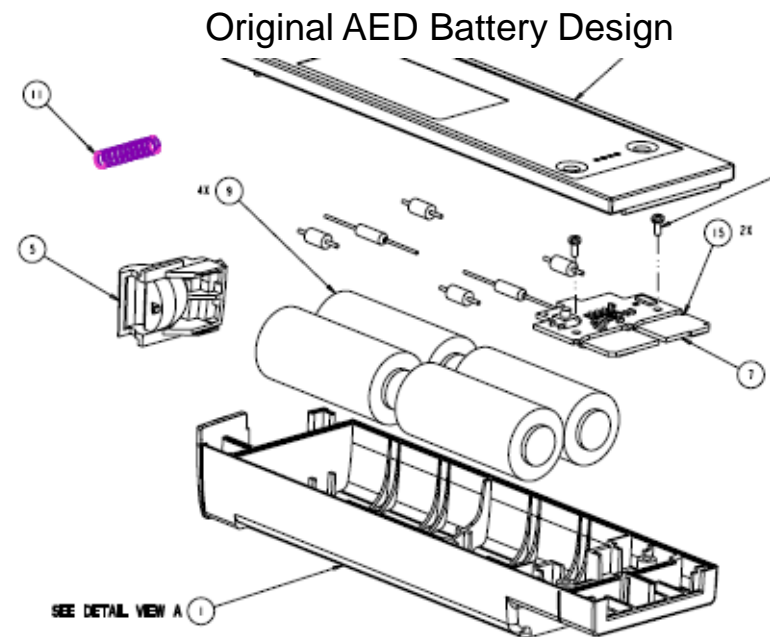


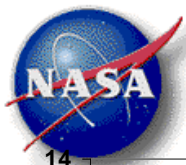
# AED LiMnO<sub>2</sub> battery

**AED for International Space Station (ISS) Human-rated Application (Flown in 2008 and has been on-orbit since)**

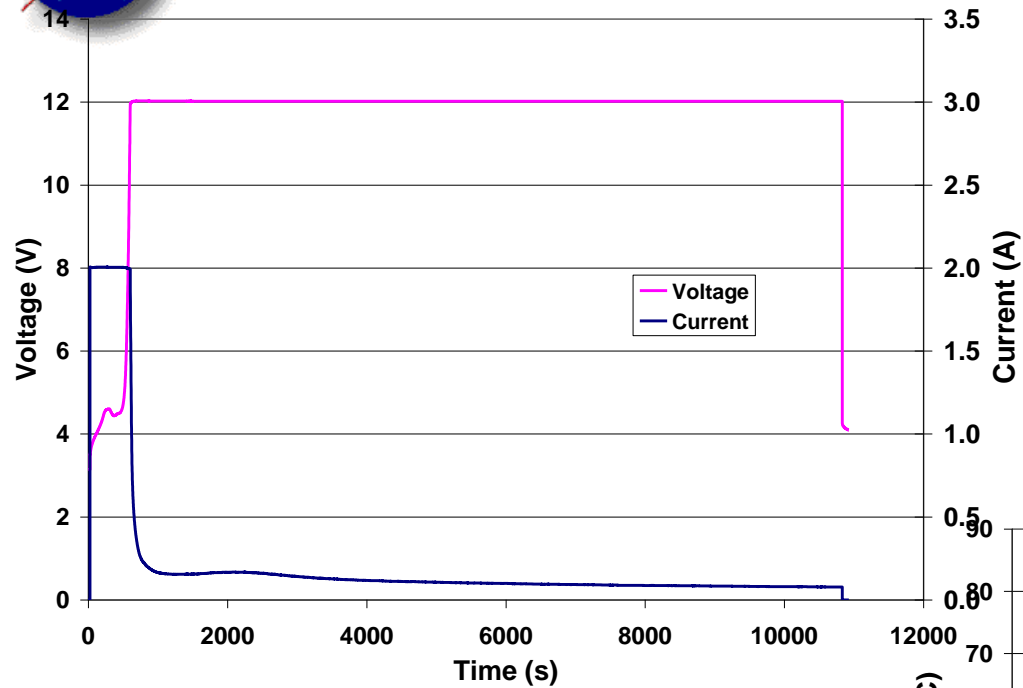
**Original Design:** Ultralife 5/4 C (U10021) high-rate cells; 4 in series; 1 diode in parallel to each cell; 2 TCOs one between cells 1 & 2 and one between cells 3 & 4

**12 V; 6.2 Ah**

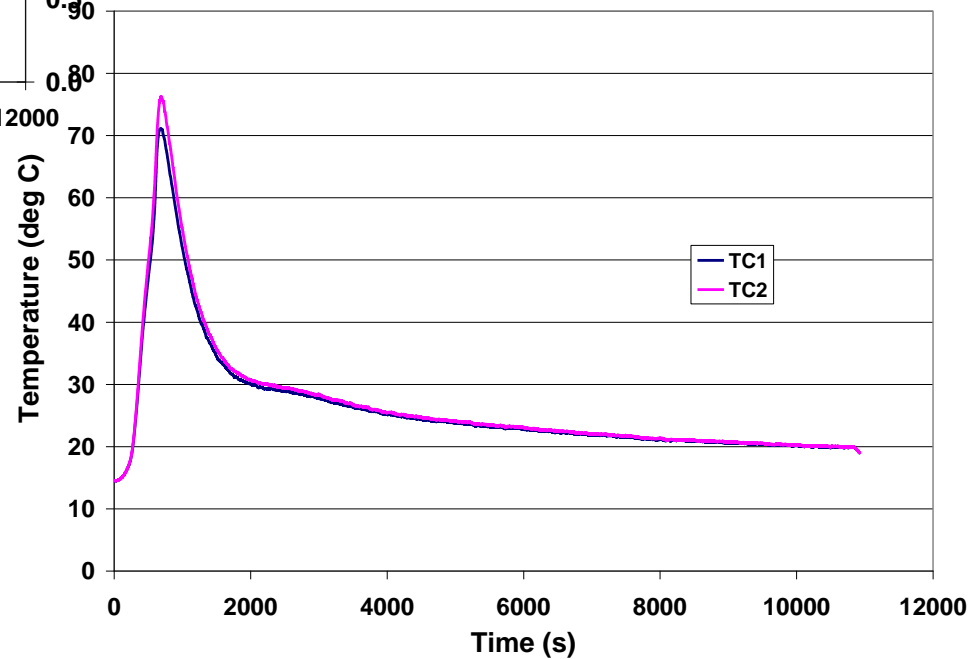


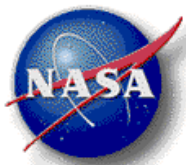


# Inadvertent Charge on Single Cell



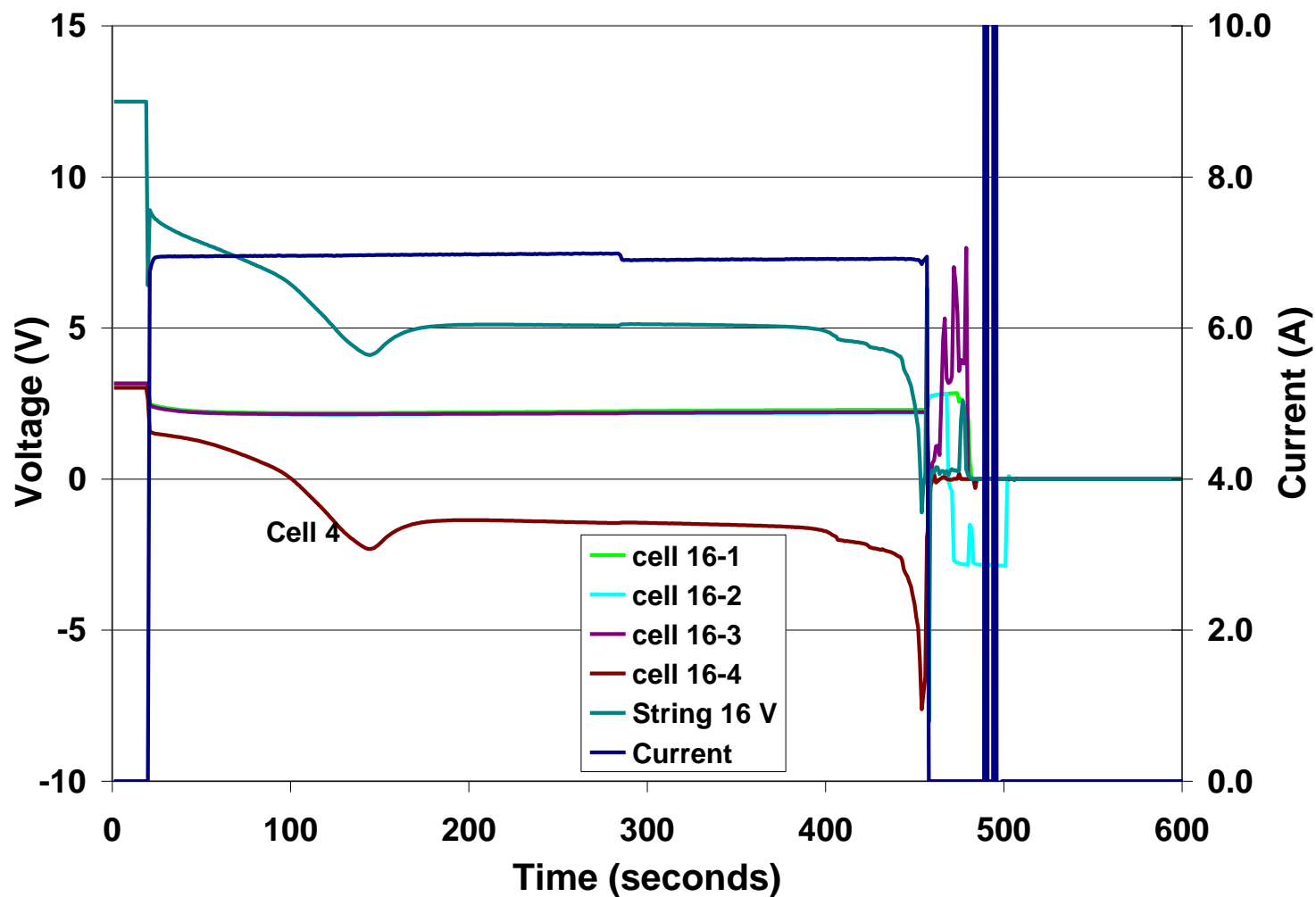
Charge Current: 2.0A  
Duration: 6 hours  
Limit: 12 V

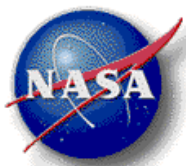




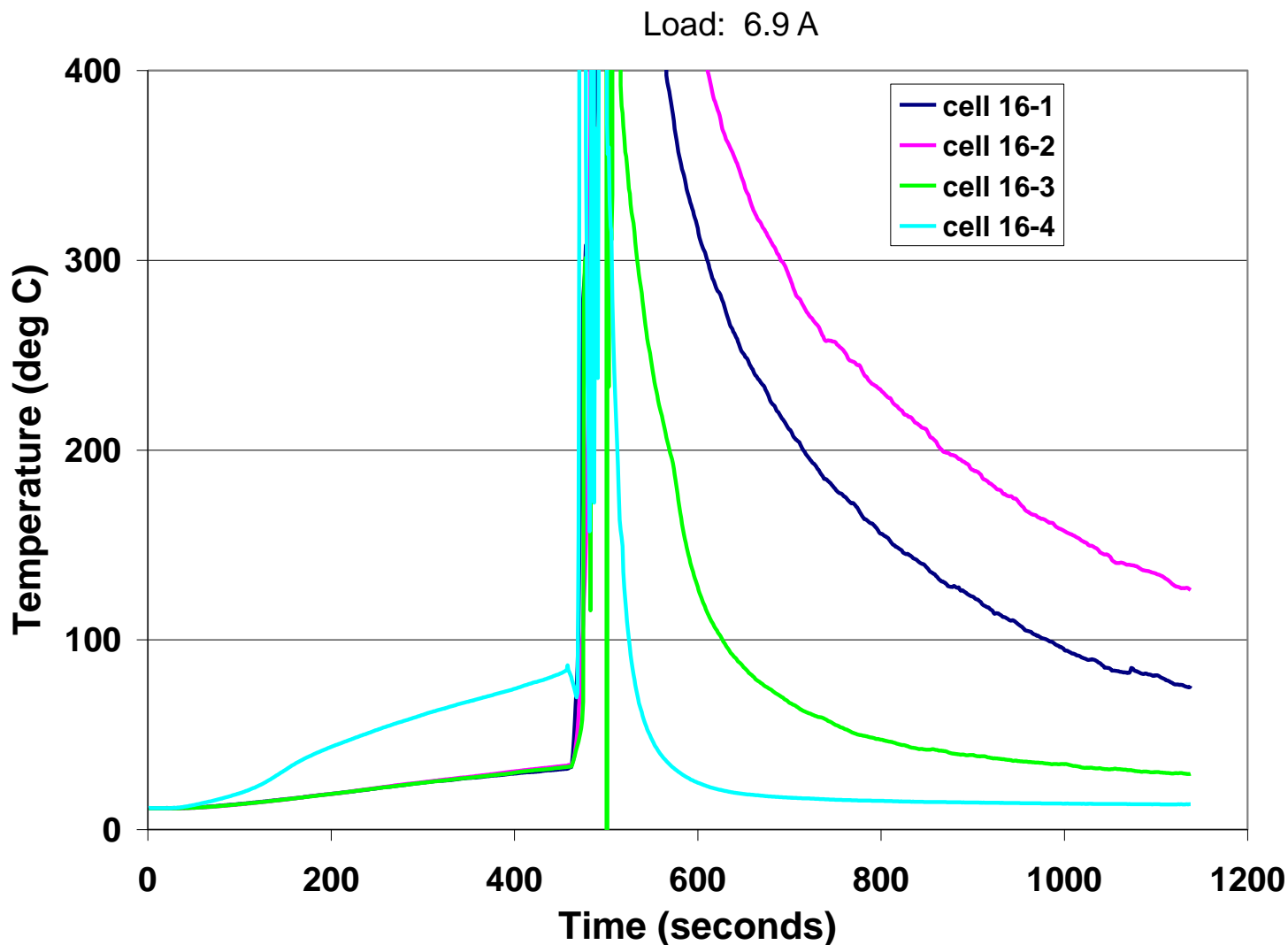
# Overdischarge of Unbalanced String of Four Cells

Load: 6.9 A

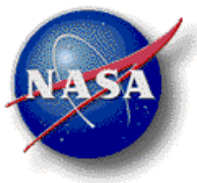




# Overdischarge of Unbalanced String of Four Cells

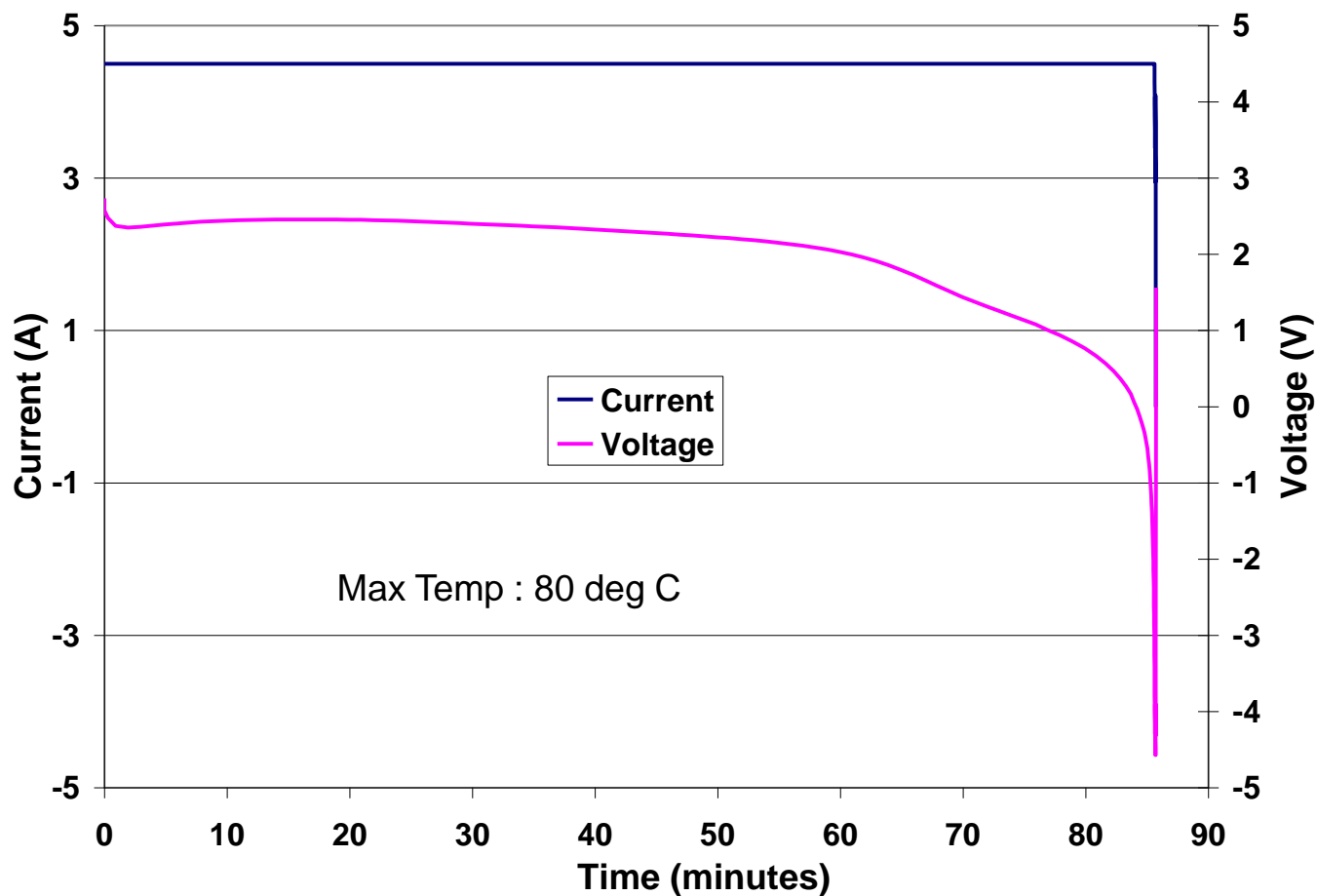


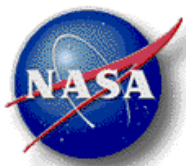




# Overdischarge Into Reversal of Single Cell

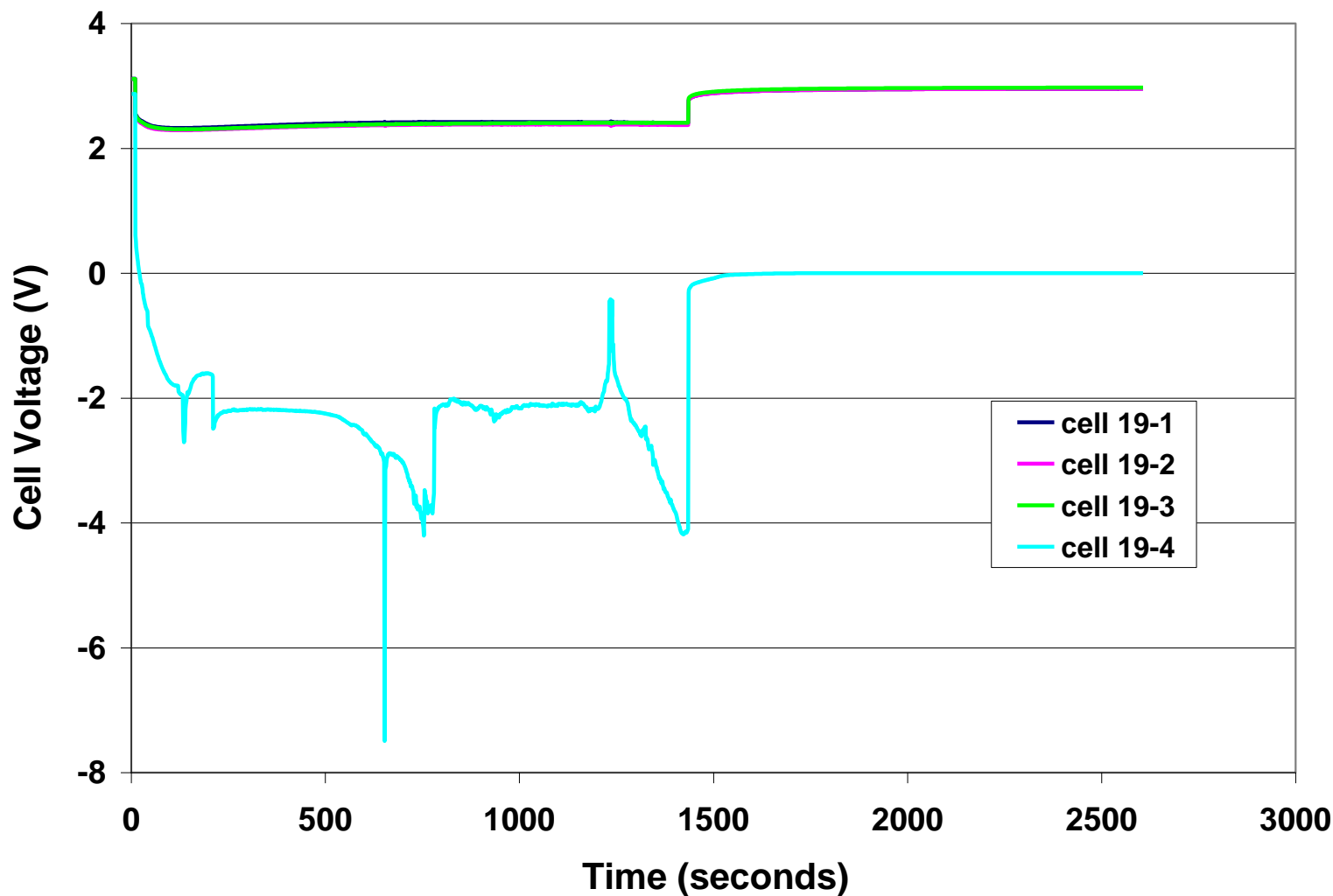
Load: 4.5 A

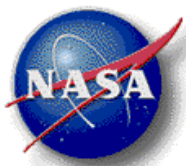




# Overdischarge of Unbalanced String of Four Cells

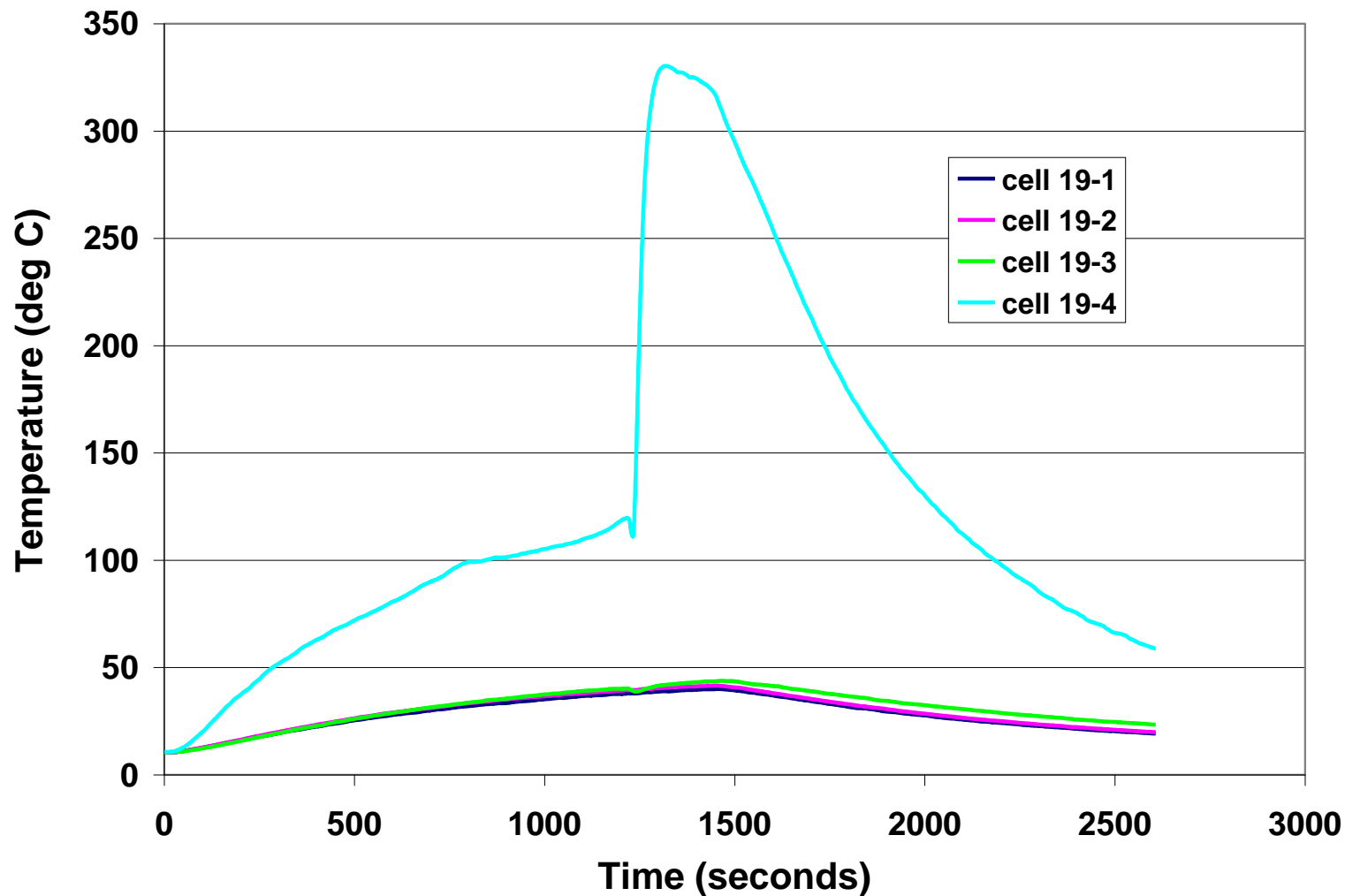
Load: 4.5 A

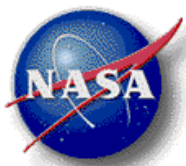




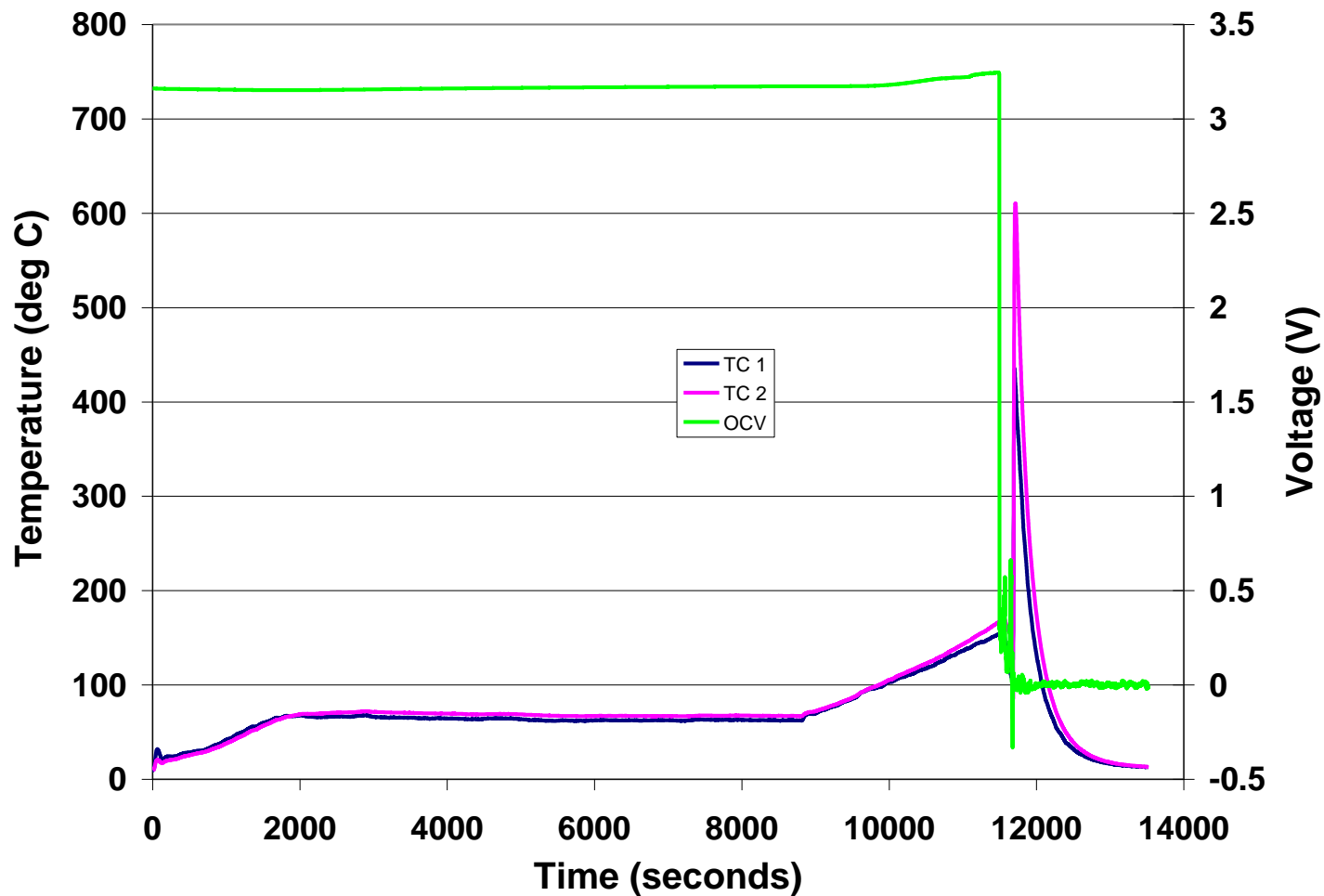
# Overdischarge of Unbalanced String of Four Cells

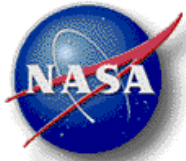
Load: 4.5 A



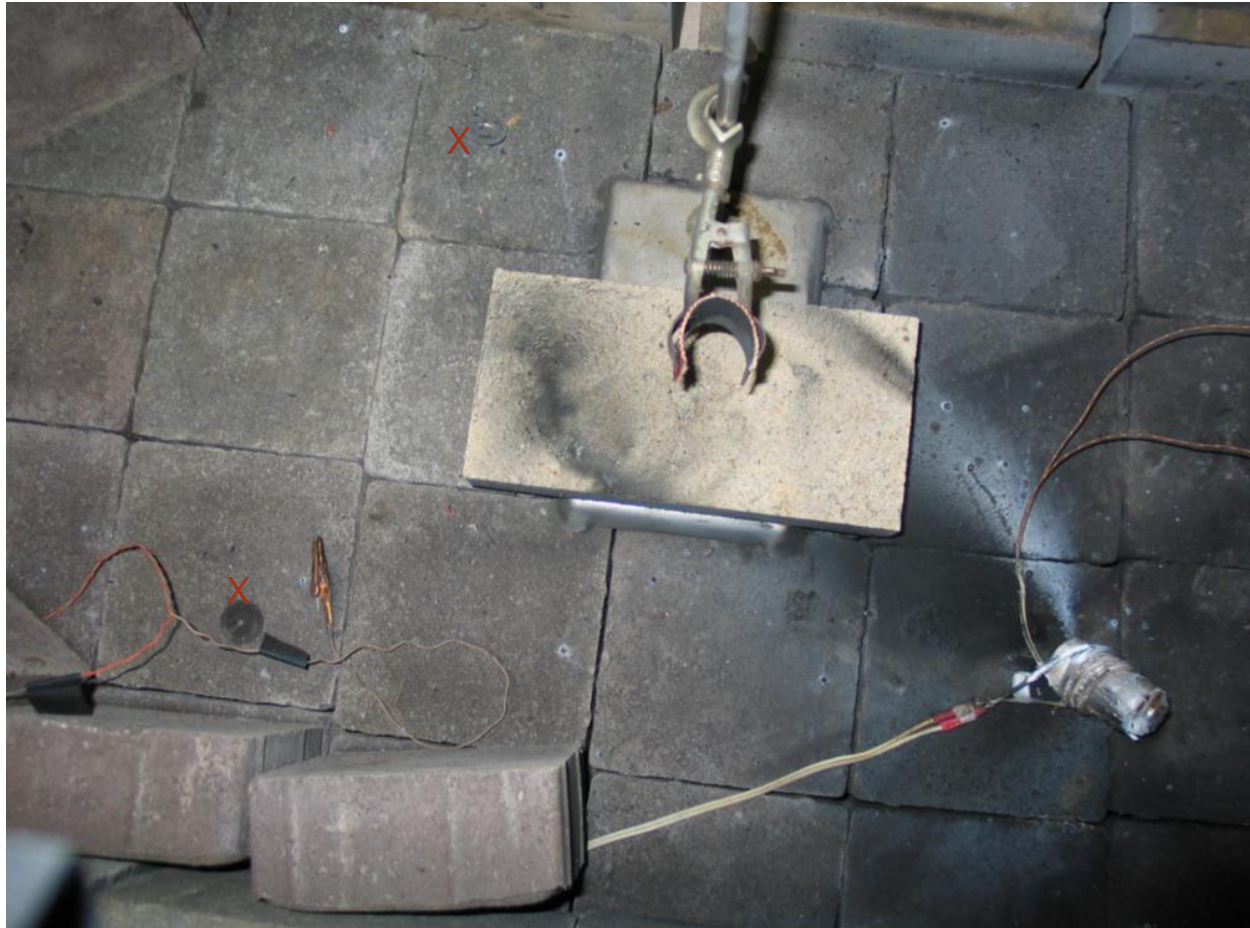


# Heat-to-Vent Test on Single $\text{LiMnO}_2$ AED Cell

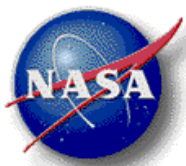




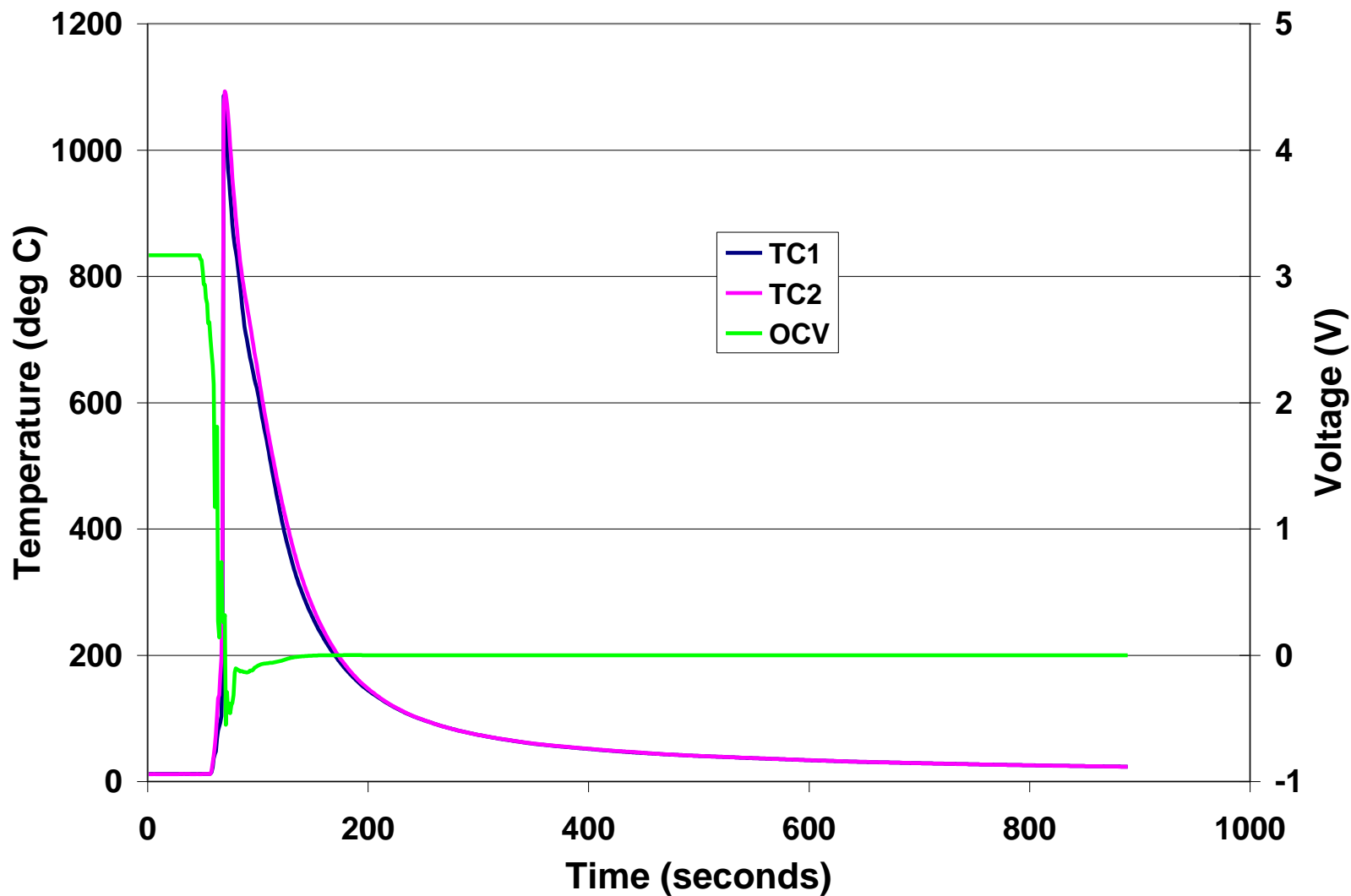
# Heat-to-Vent of Single Cell

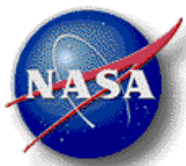


- Originally, the cell was in the clamp on the ring stand.
- the cell went to the right, the feedthrough went towards the top to the left,
- the false cover went to the bottom left,
- the brick on the lower center was in alignment with the brick next to it (the bricks are  $5 \frac{1}{2} \times 5 \frac{1}{2} \times 2 \frac{3}{8}$  in. (14 x 14 x 6 cm) and weigh 5.8 lb (2.6 kg))



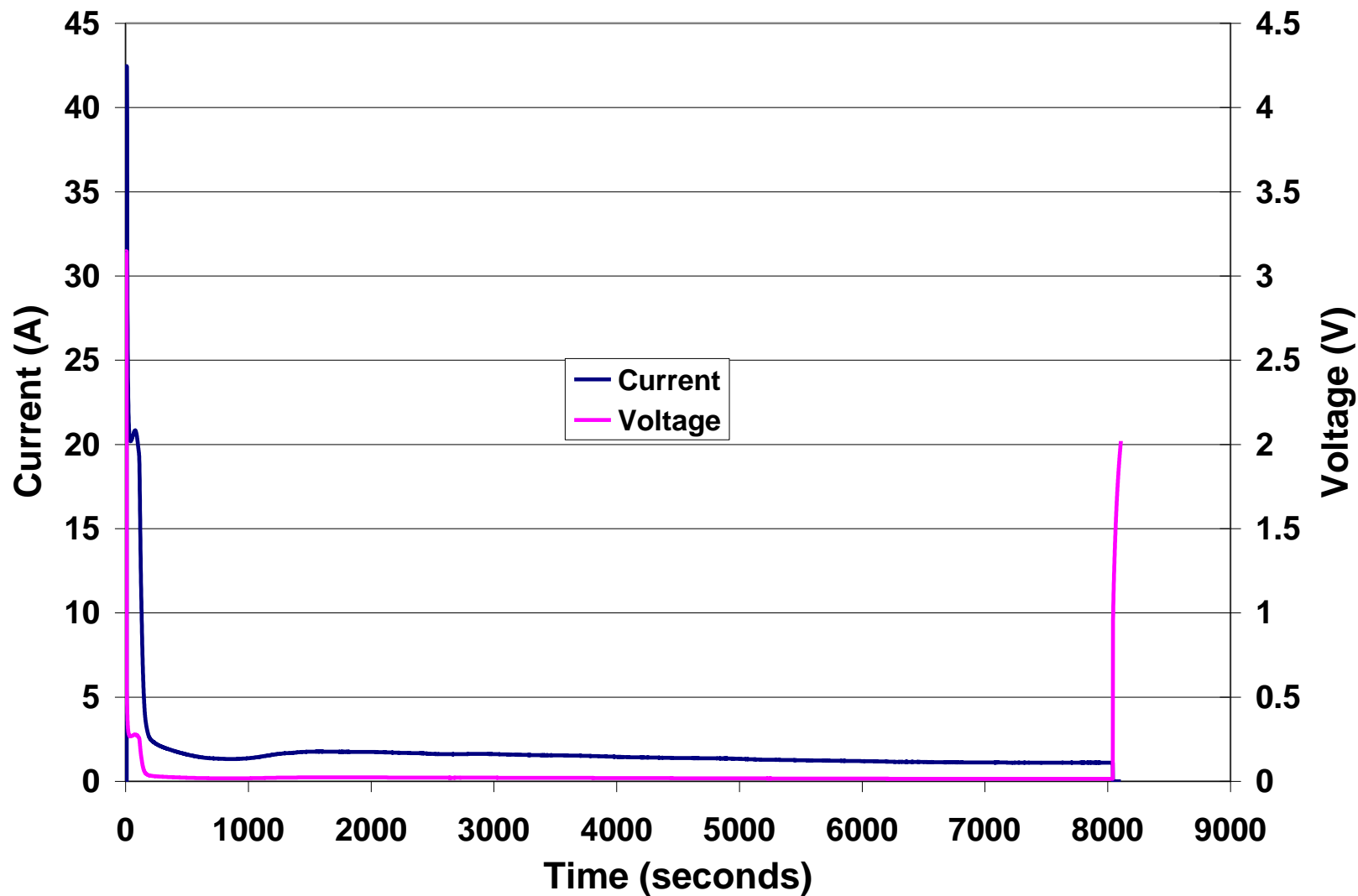
# Simulated Internal Short (Crush) on Single Cell

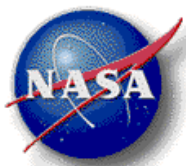




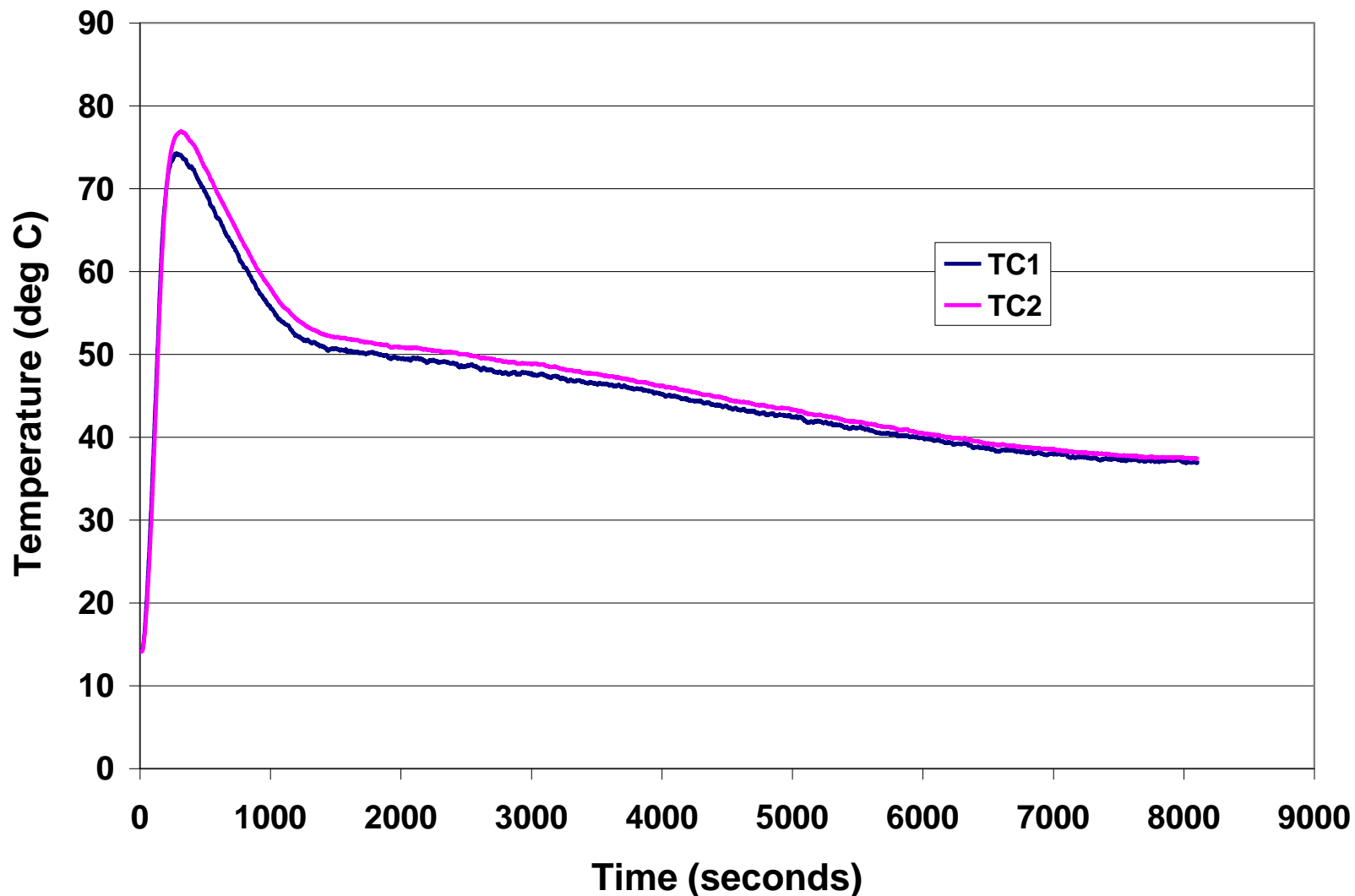
# External Short Circuit Test on a Single Cell

Load : 50 mohm

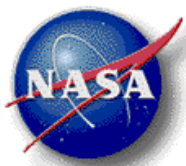




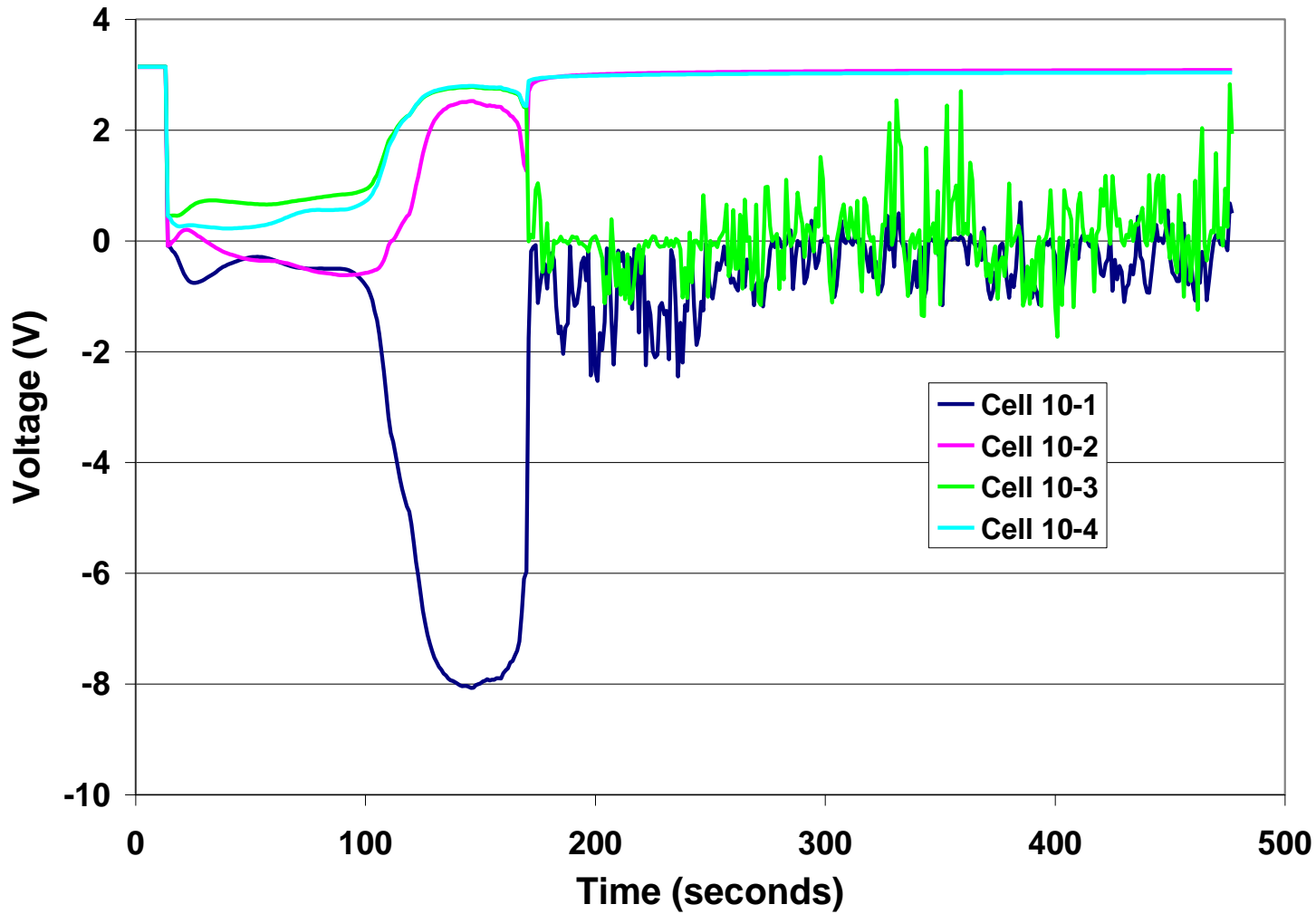
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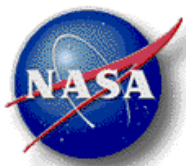




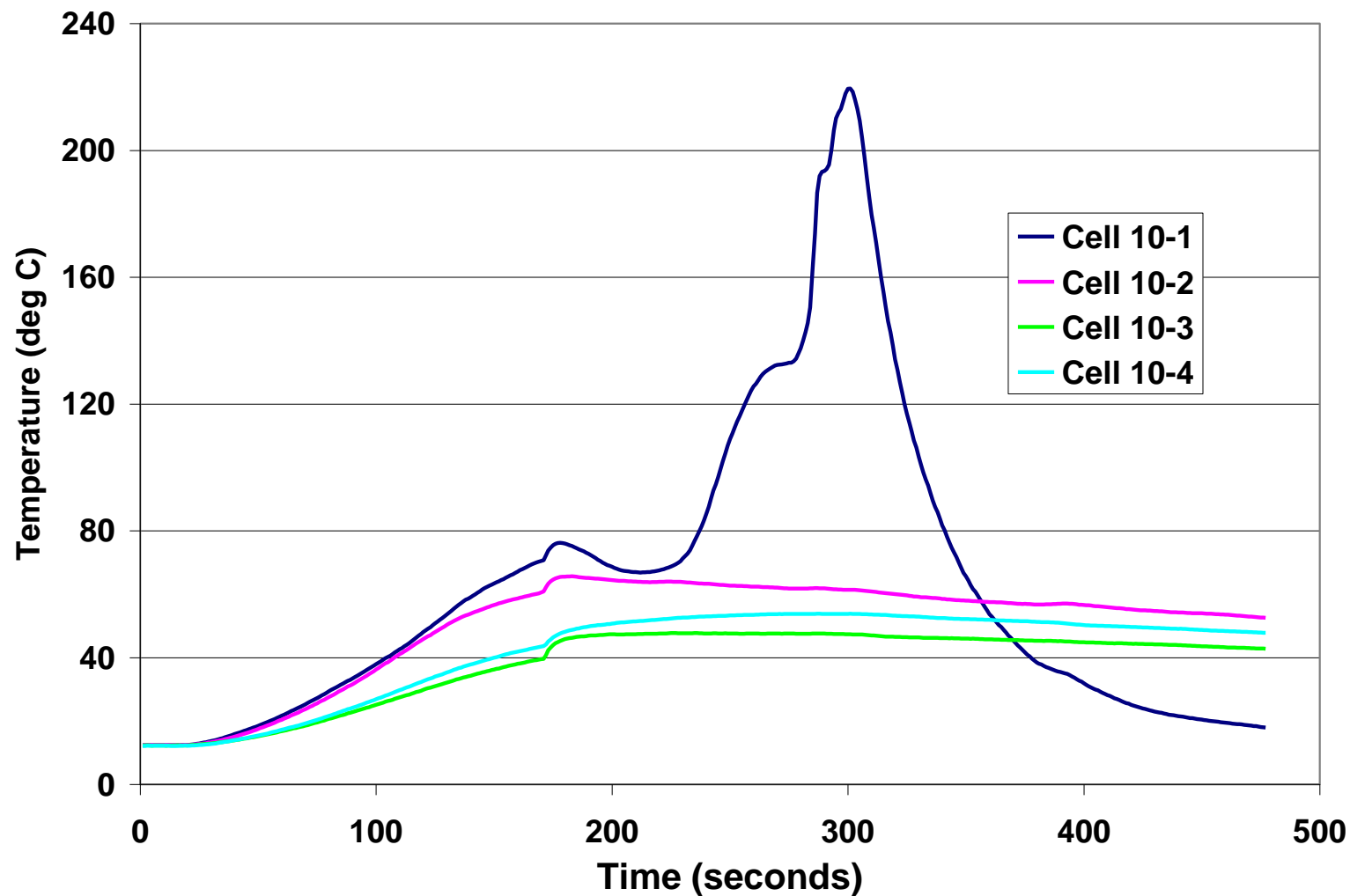


# External Short on a String of Four Cells



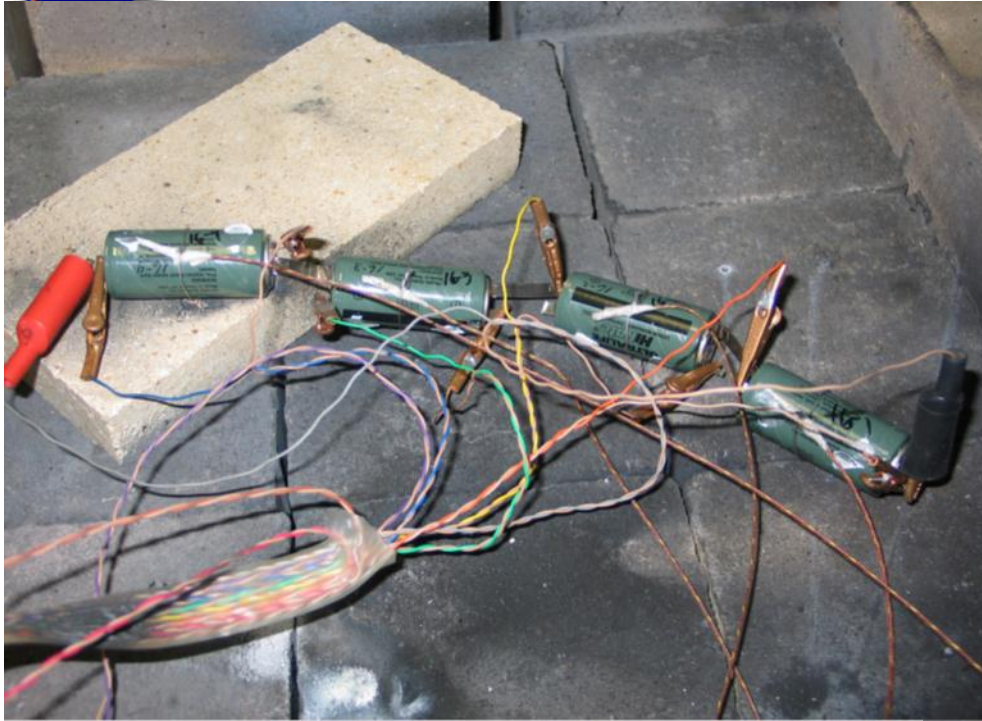


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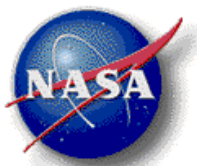




# External Short on a String of Four Cells

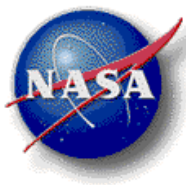


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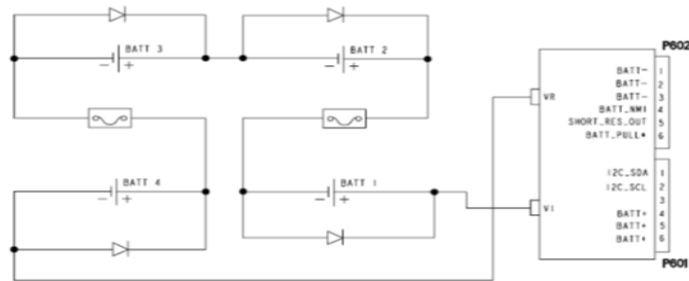
# Summary of Test Results

- Single cell tests were benign for external short, inadvertent charge and overdischarge into reversal up to 4.5 A.
  - At lower current loads cells die (may be due to excessive dendrite formation) benignly.
- String level external short circuits lead to an unbalanced overdischarge, with one cell going into reversal. The result is catastrophic violent venting.
- Unbalanced string overdischarges at different currents causes catastrophic violent venting also.
- Heat-to-vent is very dramatic displaying violent venting
- Simulated internal short is also catastrophic and displays violent venting.
- Battery is not UL-rated; hence does not have dual-fault tolerance or tolerance to inherent cell tolerance to failures

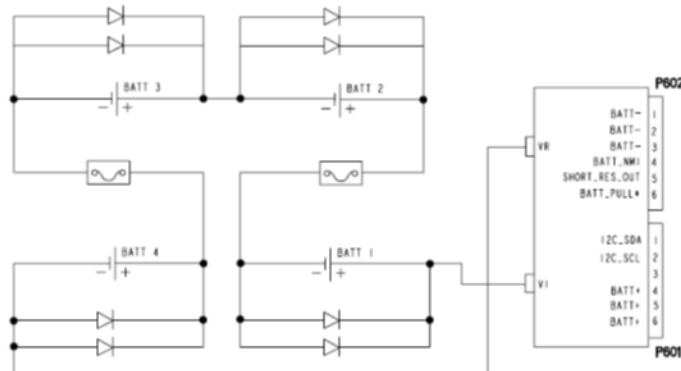


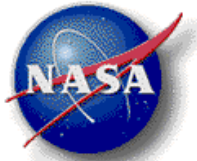
# Design Change as Recommended by NASA Battery Safety

Original AED Battery Design



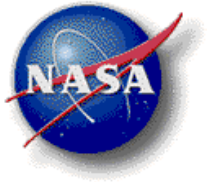
NASA JSC AED Battery Design (currently on ISS)





# Summary and Conclusions

- Single cell tests were benign for external short, inadvertent charge and overdischarge into reversal up to 4.5 A.
  - At lower current loads cells die (may be due to excessive dendrite formation) benignly.
- String level external short circuits lead to an unbalanced overdischarge, with one cell going into reversal. The result is catastrophic violent venting.
- Unbalanced string overdischarges at different currents causes catastrophic violent venting also.
- Heat-to-vent is very dramatic displaying violent venting
- Simulated internal short is also catastrophic and displays violent venting.
- Battery is not UL-rated; hence does not have dual-fault tolerance or tolerance to inherent cell tolerance to failures
- Battery Design for NASA JSC's human-rated application for use on ISS was changed to include two bypass diodes per cell to provide for two-failure tolerance to overdischarge into reversal (and external short) hazards.



# Acknowledgment

- Applied Power International
- Space and Life Sciences Directorate, NASA-JSC